## Amendments to the Claims:

This listing of claims will replace all prior versions, and listings, of claims in the application:

## Listing of Claims:

1 - 18. (cancelled)

19. (currently amended) A substrate having at least one surface and a zinc diffused nickel alloy coating on said at least one surface, said coating having a first layer formed from nickel or a nickel alloy applied to said at least one surface and a second layer formed from zinc applied over said first layer, said first layer having zinc atoms diffused therein, and said substrate being formed from a deoxidized low carbon steel.

## 20. (cancelled)

- 21. (original) A substrate according to claim 19, wherein said substrate comprises a component used in a gas turbine engine.
- 22. (currently amended) A substrate according to claim 19, wherein said zinc diffused nickel alloy coating provides corrosion resistance and heat resistance at temperatures in excess of 900°F.
- 23. (currently amended) A substrate according to claim 19, wherein said <del>coating has a nickel or nickel alloy layer into which zine atoms have diffused and a zine layer into which</del> second layer has nickel atoms <del>have</del> diffused therein.

24. (currently amended) A substrate according to claim 23, wherein said <u>first layer is a nickel alloy layer is</u> formed by an alloy selected from the group consisting of a nickel cobalt alloy, a nickel iron alloy, a nickel manganese alloy, a nickel molybdenum alloy, and a nickel tin alloy.

## 25. (cancelled)

- 26. (currently amended) A component for use in a gas turbine engine comprising:
  - a steel substrate formed from a low carbon steel; and
- a zinc diffused nickel alloy coating on said steel substrate, said coating having a first layer formed from a nickel alloy deposited on and in contact with a surface of said substrate and a second layer formed from zinc deposited on said first layer and said first layer having zinc atoms diffused therein.
- 27. (currently amended) A method for forming a corrosion and heat protective coating on a gas turbine engine component comprising the steps of:

providing a substrate formed from a low carbon steel;

forming a nickel base alloy coating layer on said substrate;

applying a layer of zinc over said nickel base alloy coating layer; and

diffusing the zinc into said nickel  $\frac{\text{base}}{\text{base}}$  alloy coating layer.

- 28. (new) A component according to claim 26, wherein said nickel alloy layer is formed from a nickel cobalt alloy.
- 29. (new) A component according to claim 26, wherein said nickel alloy layer is formed from a nickel iron alloy.
- 30. (new) A component according to claim 26, wherein said nickel alloy layer is formed from a nickel manganese alloy.
- 31. (new) A component according to claim 26, wherein said nickel alloy layer is formed from a nickel molybdenum alloy.
- 32. (new) A component according to claim 26, wherein said nickel alloy layer is formed from a nickel tin alloy.
- 33. (new) A method according to claim 27, wherein said nickel alloy coating layer forming step comprises depositing a nickel cobalt alloy on said substrate.
- 34. (new) A method according to claim 27, wherein said nickel alloy coating layer forming step comprises depositing a nickel iron alloy on said substrate.
- 35. (new) A method according to claim 27, wherein said nickel alloy coating layer forming step comprises depositing a nickel manganese alloy on said substrate.

- 36. (new) A method according to claim 27, wherein said nickel alloy coating layer forming step comprises depositing a nickel molybdenum alloy on said substrate.
- 37. (new) A method according to claim 27, wherein said nickel alloy coating layer forming step comprises depositing a nickel tin alloy on said substrate.